The AMMS is a powerful computing system designed to provide processing features to rotary wing aircraft, mainly:

- Navigation & Flight Management,
- Cockpit Displays Management,
- Aircraft Plants Management.

The System is composed by:

- Two AMMCs Units (the Computer Core)
- Two Data Transfer Units
- A Control Panel

The AMMS provides the capabilities to acquire, process and manage information and aircraft plant data from EFIS/EICAS, relevant CNI equipment and AFCS, in a reliable, high performance environment.

The AMMS is based upon a core consisting of two identical computers (AMMC, Aircraft and Mission Management Computers), synchronised and concurrently performing the same operations in a master/hot-standby architecture, which allows, in case of failure of the Master, a totally automatic reconfiguration for the all system functionalities, therefore resulting into an high mission availability.

KEY FEATURES
- Redundant LRI configuration;
- Power PC G4 MCS-E 500Mhz processor;
- Easy add on of additional processor module and I/O;
- I/O configuration for different aircraft;
- COTS RTOS based on I-178B by Green Hills;
- Dedicated equipment software and ADA development environment for application;
- Digital maps: symbol generation & vectorial layers;
- Synthetic voice generator;
- Redundant MIL-STD 1553B I/F operates as BC/BM/RT;
- High communication throughput (Ethernet 10/100 baseT, MIL-STD-1553b, ARINC 429, ARINC 739, RS485 HDLC);
- AFDX Copper/Optical I/F compliant with ARINC664;
- High data storage capability (up to 8GB);
- Civil specification compliance: RTCA/DO-160D, RTCA/DO-254, RTCA/DO178B.
FUNCTIONALITIES

Flight Management System (FMS)
- Area Navigation Management
- Tactical Patterns Management
- Navigation Database Management
- Navigation Display data management
- Helicopter and Navigation Performance computation
- Horizontal and Vertical Steering commands generation for AFCS

Data Control and Equipment Management/Preset
- Mission Equipment
- Communication/Identification Equipment
- Navigation Sensors

Aircraft Systems Interfacing, Data Acquisition, Status monitoring and Alert generation
- Engines - Rotors and Transmission - Hydraulics - Electrical - Fuel

Helicopter plants Health and Usage Monitoring and Maintenance (HUMS)
- Vibration signal acquisition and processing for Transmission Monitoring
- Engines Health and Usage Monitoring
- Structural Usage Monitoring

Display Management for EICAS and Navigation information
- Helicopter Data Upload/Download - Navigation Data - COMMS Data - Map Data - Mission Data - HUMS Data - Maintenance Data

TECHNICAL SPECIFICATIONS

AMMC is a 1 ATR size housing, with integrated power supply and up to 13 MCS standard boards.
- Processor G4 MCS-E Power PC 500 MHz, 256 MB RAM, 96 MB Flash, 2 MB cache, with a 2 MB NOVRAM.
- Synthetic Voice Generator
- Digital Map Generator: G4 MCS-E with graphic mezzanine for map generation functionality

Interfaces
- 2 MIL-STD-1553 BUS I/F BC/BM/RT
- 20 TX ARINC 429 and 34 RX ARINC 429 channels
- 4 TX and 4 RX ARINC 429 Non-Time-Framed channels
- 250 Discrete Input Signals
- 17 Discrete Output Signals
- 6 AC Analogue High Level
- 54 DC Analogue Input
- 17 Accelerometers sensors
- Magnetic Pick-Up
- STANAG 3350 RGB Video output
- Synthetic Voice Generator

Technical Features (AMMC)
- Size: 1 ATR Short
- Weight: 15.2 Kg (bi-processor)
- Power requirements: 115 VAC @ 400 Hz
- Power consumption: 180 W
- Cooling: Forced air cooling (ARINC 600)
- MTBF: 2500 operating hours

Control Panel
- Control Panel provides the crew with current system status information and allows manual mastership selection.

Data Transfer Units (DTU)
- The Data Transfer Units, connected to AMMCs via Fast Ethernet interface, allow data download and upload on PCMCIA Type II media (up to 8 GB).

The AMMS is the Mission Management System for the AgustaWestland EH101 helicopter.